# **MMOD-IMLI: Integrated Thermal Insulation and Micrometeoroid/Orbital Debris Protection**

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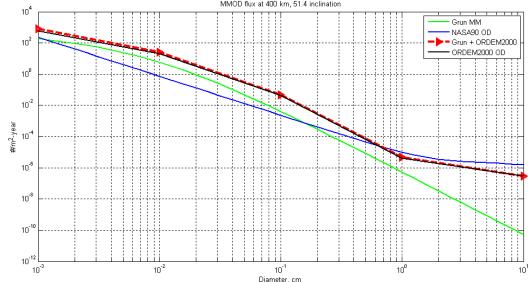
Proprietary, Patented and Patent Pending Technology of **Quest Thermal Group and Ball Aerospace** 

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#### **MMOD** Protection



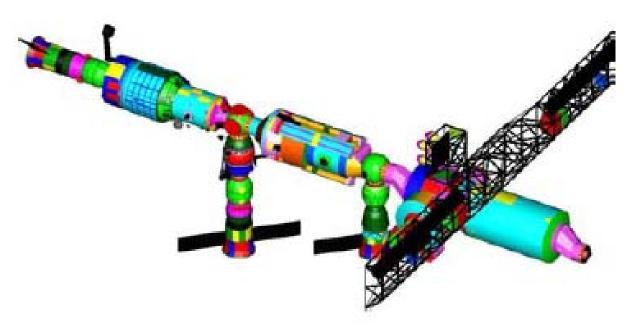
Micrometeoroid/Orbital Debris (MMOD) is a risk for spacecraft, fuel depots and space stations
Designers must provide MMOD protection, based on MMOD environment, size/geometry/orientation of spacecraft, duration of mission, and likelihood of critical damage



# MMOD shielding on the ISS



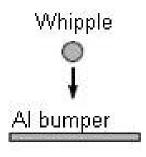
- Adequate MMOD protection on the ISS is required for crew safety and mission success
- The ISS requires multiple MMOD shields
- Shield design based on risk assessments

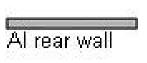


### MMOD shields



- Whipple shields
  - Two plate shields
  - Spacing between outer bumper layer and spacecraft rear wall is critical to stopping power
  - To stop a 6.3mm particle @ 7km/s requires 20 kg/m<sup>2</sup>

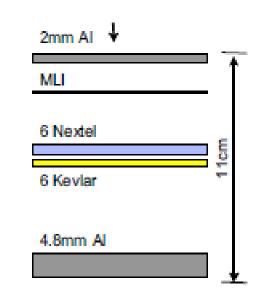




#### Quest Thermal Group and Ball Aerospace Information

### **MMOD** shields

- Stuffed Whipple shields
  - Uses additional high strength layers
  - Protects US Lab module of ISS
  - Designed to stop 1.3cm @ 7km/s
  - Massive at 27kg/m<sup>2</sup> (42kg/m<sup>2</sup> with rear wall)





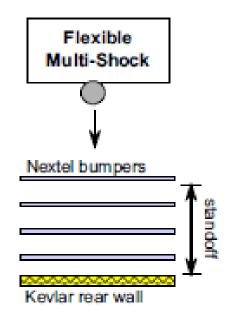
Quest Thermal Group and Ball Aerospace Information

# MMOD shields

#### Nextel Multi-shock multiple layer shields

- 4 layers Nextel ceramic fiber, Kevlar polyaramid layer
- Penetration equations developed
- Stopping power related to spacing and areal density

Areal Density <sub>bumpers</sub> =	0.19 * diameter <sub>particle</sub> * density <sub>particle</sub>					
Areal Density <sub>rear wall</sub> = -	43.1 * Mass <sub>particle</sub> * Velocity <sub>particle</sub>					
	Spacing <sup>2</sup> * (40/Yield Stress <sub>rear wall</sub> ) <sup>0.5</sup>					

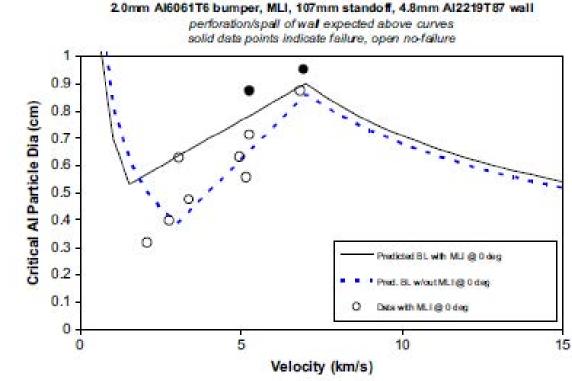




#### **MMOD** shields



#### Conventional MLI offers slight MMOD protection

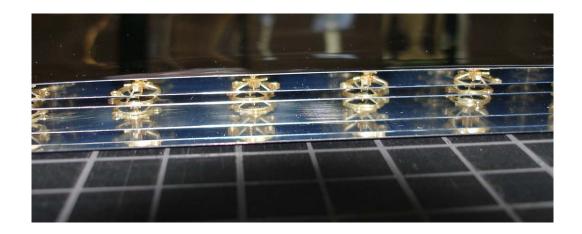


## **IMLI discrete spacers**



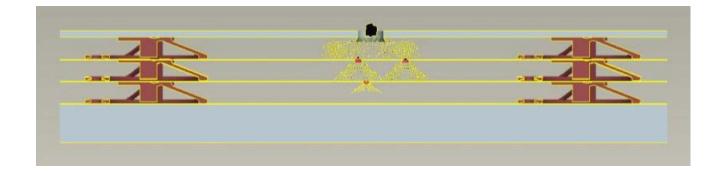
#### IMLI discrete spacer technology

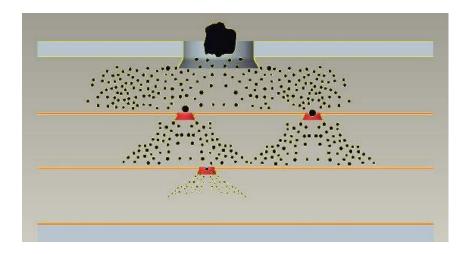
- Can provide large interlayer spacing
- Can support high strength ballistic layers
- Can provide high performance thermal insulation
- Heavy bumper plate and standoffs not required



#### **MMOD-IMLI concept**



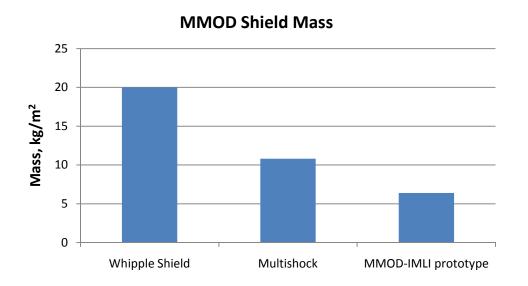








 Preliminary analysis shows mass to stop penetration by a 6.3mm particle at 7km/s:



# **MMOD-IMLI** Phase I goals



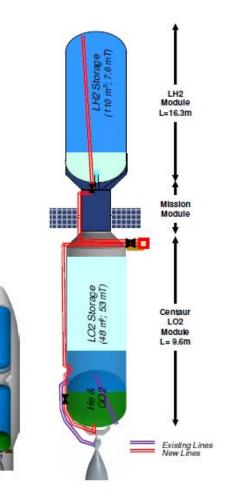
- Evaluate MMOD protection and thermal performance available from MMOD-IMLI structures
- Design and fabricate MMOD-IMLI prototypes
- Perform hypervelocity impact tests
- Measure thermal performance
- Determine feasibility of MMOD-IMLI

# **Orbital Fuel Depot mission**

#### Orbital Fuel Depot (OFD)

- Requires near zero boiloff (thermal insulation)
- 15 year mission
- Colocated near ISS at 51.6°, 400km
- Acceptable risk of failure set at 5% (95% PNP)

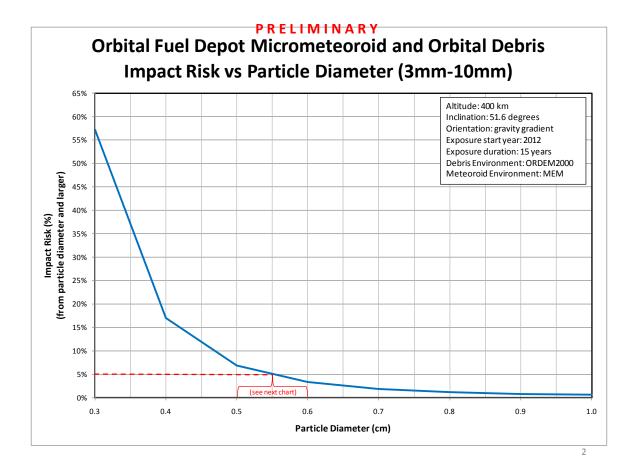






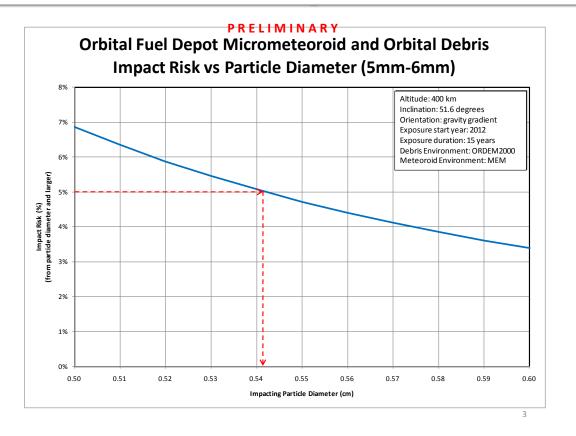
#### MMOD fluence and risk





#### **MMOD fluence and risk**

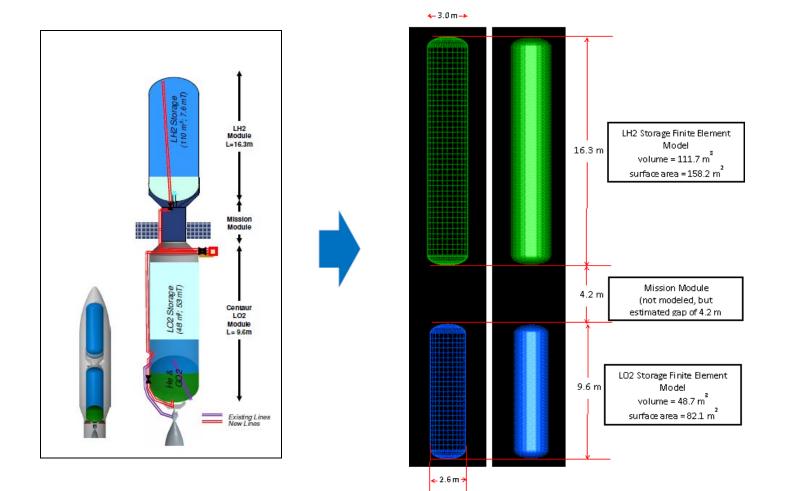




For PNP > 95%, critical particle is 5.4mm diameter.

#### **Orbital Fuel Depot FEM**

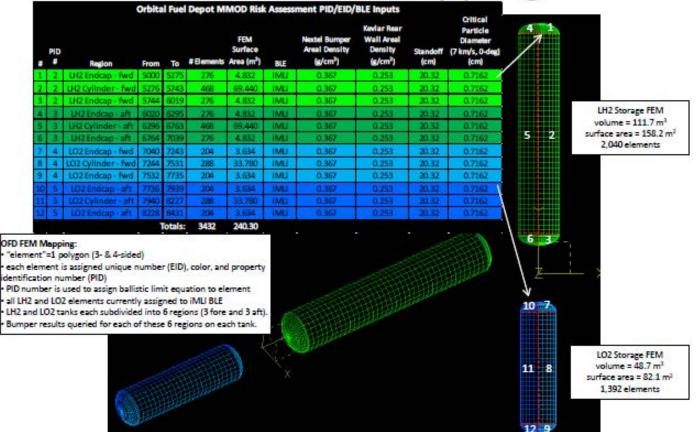




### **Orbital Fuel Depot FEM**



### **OFD FEM Mapping**

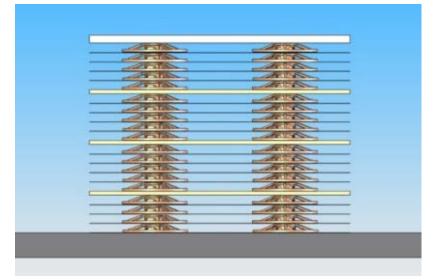


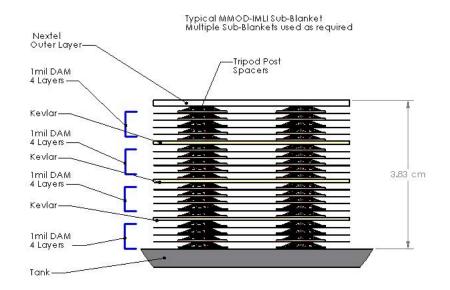
D.M. Lear NASA JSC/KX Hypervelocity Impact Technology (HVIT) Team

#### PRELIMINARY

# **MMOD-IMLI** preliminary design



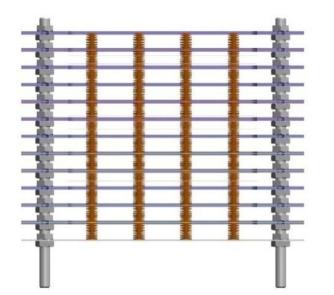




- MMOD-IMLI preliminary design:
  - Widely spaced layers
  - Supported 12 ballistic layers of Nextel and Kevlar
  - Integrated MMOD shielding and thermal insulation

### **MMOD-IMLI Areal Density**





iMLI Areal Density Estimate (g/cm <sup>2</sup> )								
Laver								
		Material	Areal					
Laver				Conversion	Density			
#	Location	Description	Factor	$(g/cm^2)$				
1	1	Nextel AF-10	(g/cm <sup>2</sup> ) 0.0292	# Layers	1.000	0.0292		
2	1.2	Mylar	0.003556	9	1.000	0.032004		
3	2	Nextel AF-10	0.0292	1	1.000	0.0292		
4	2,3	Mylar	0.003556	9	1.000	0.032004		
5	3	Nextel AF-10	0.0292	1	1.000	0.0292		
6	3,4	Mylar	0.003556	9	1.000	0.032004		
7	4	Nextel AF-10	0.0292	1	1.000	0.0292		
8	4,5	Mylar	0.003556	9	1.000	0.032004		
9	5	Nextel AF-10	0.0292	1	1.000	0.0292		
10	5,6	Mylar	0.003556	9	1.000	0.032004		
11	6	Nextel AF-10	0.0292	1	1.000	0.0292		
12	6,7	Mylar	0.003556	9	1.000	0.032004		
13	7	Kevlar KM2-705	0.0244	1	1.000	0.0244		
14	7,8	Mylar	0.003556	9	0.667	0.021336		
15	8	Kevlar KM2-705	0.0244	1	1.000	0.0244		
16	8,9	Mylar	0.003556	9	0.667	0.021336		
17	9	Kevlar KM2-705	0.0244	1	1.000	0.0244		
18	9,10	Mylar	0.003556	9	0.667	0.021336		
19	10	Kevlar KM2-705	0.0244	1	1.000	0.0244		
20	10,11	Mylar	0.003556	9	0.667	0.021336		
21	11	Kevlar KM2-705	0.0244	1	1.000	0.0244		
22	11,12	Mylar	0.003556	9	0.667	0.021336		
23	12	Kevlar KM2-705	0.0244	1	1.000	0.0244		
24	13	Mylar	0.003556	9	0.000	0		
000000		AI 2024-T3				100		
25	WP	(0.040")		0	0.000	0		
					real Density:			
Rearwall Areal Density: 0.253								
	Total Shield Areal Density: 0.620							

#### **OFD Penetration Risk**

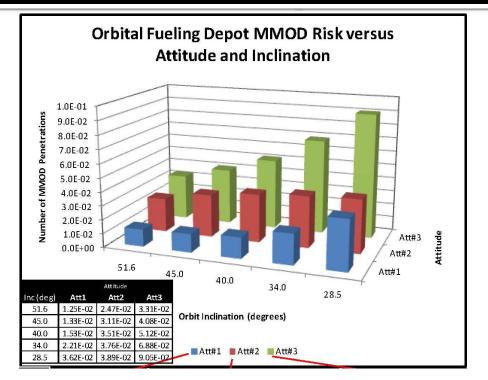


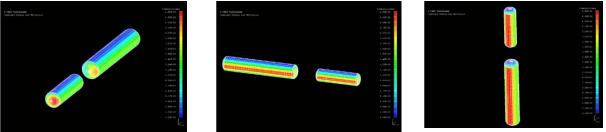
	Orbital Fuel Depot MMOD Number of Penetrations (N)											
	(Inclination 51.6 deg, 2012+15 years exposure, 400 km altitude, ORDEM2000 & MEM)											
			Attitude #1 (RPY=0,0		.0,0)	0,0) Attitude #2 (RPY = 0,0,90)			Attitude #3 (RPY = -90,0,0)			
#	Region	From	То	deb	met	both	deb	met	both	deb	met	both
1	LH2 Endcap - fwd	5000	5275	8.57E-04	2.44E-05	8.82E-04	1.48E-04	8.58E-06	1.56E-04	3.56E-04	9.39E-06	3.65E-04
2	LH2 Cylinder - fwd	5276	5743	4.80E-03	4.17E-04	5.21E-03	7.09E-03	2.23E-04	7.31E-03	1.42E-02	4.65E-04	1.47E-02
3	LH2 Endcap - fwd	5744	6019	7.33E-04	2.33E-05	7.57E-04	6.38E-04	3.02E-05	6.68E-04	3.56E-04	3.35E-05	3.89E-04
4	LH2 Endcap - aft	6020	6295	5.26E-04	1.24E-05	5.38E-04	1.49E-04	8.52E-06	1.58E-04	1.33E-04	3.48E-06	1.37E-04
5	LH2 Cylinder - aft	6296	6763	1.74E-04	1.17E-04	2.91E-04	7.10E-03	2.20E-04	7.32E-03	6.16E-03	1.47E-04	6.31E-03
6	LH2 Endcap - aft	6764	7039	3.18E-04	1.12E-05	3.29E-04	6.37E-04	3.00E-05	6.67E-04	1.33E-04	1.64E-05	1.49E-04
7	LO2 Endcap - fwd	7040	7243	5.16E-04	1.75E-05	5.34E-04	1.09E-04	6.13E-06	1.15E-04	2.70E-04	7.02E-06	2.77E-04
8	LO2 Cylinder - fwd	7244	7531	2.33E-03	2.03E-04	2.54E-03	3.45E-03	1.08E-04	3.56E-03	6.92E-03	2.26E-04	7.15E-03
9	LO2 Endcap - fwd	7532	7735	6.41E-04	1.82E-05	6.59E-04	4.93E-04	2.49E-05	5.18E-04	2.70E-04	2.59E-05	2.95E-04
10	LO2 Endcap - aft	7736	7939	1.86E-04	8.33E-06	1.94E-04	1.10E-04	6.08E-06	1.16E-04	1.01E-04	2.60E-06	1.03E-04
11	LO2 Cylinder - aft	<b>79</b> 40	8227	8.49E-05	5.68E-05	1.42E-04	3.45E-03	1.07E-04	3.56E-03	3.00E-03	7.18E-05	3.07E-03
12	LO2 Endcap - aft	8228	8431	3.97E-04	9.15E-06	4.06E-04	4.91E-04	2.48E-05	5.16E-04	1.01E-04	1.31E-05	1.14E-04
Ţ	fotal Number of Pene	tratio	ıs (N):	1.16E-02	9.18E-04	1.25E-02	2.39E-02	7.97E-04	2.47E-02	3.20E-02	1.02E-03	3.31E-02
			7		<b>PNP</b>	0.988		PNP	0.976		PNP	0.967
		8		N	PNPreq	0.950		PNP <sub>req</sub>	0.950		PNP <sub>req</sub>	0.950
		_	PNP=	e n	N/N <sub>req</sub>			N/N <sub>req</sub>	0.481		N/N <sub>req</sub>	0.644

- Analysis predicts with 120-layer MMOD-IMLI:
  - For 100 m<sup>2</sup> OFD, 51.6° 400km orbit, 15 year mission:
  - There would be 0.0125 penetrations
  - PNP = 98.8% or 24% of the allowed risk

### **OFD Penetration Risk**







# Hypervelocity Impact Testing

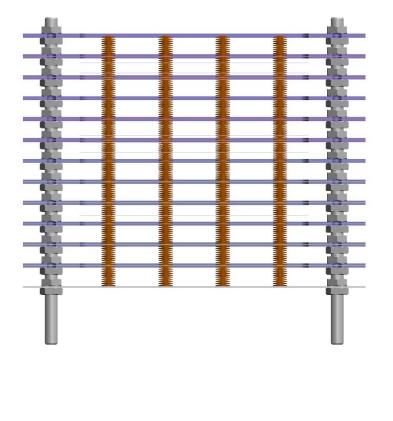


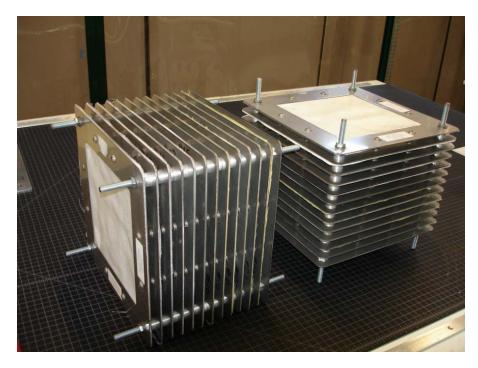


- Quest deemed HVI testing critical:
  - Quest paid for HVI shots at White Sands Test Facility
  - Two shots were planned, to give two data points for preliminary BLE development
  - One shot was an Al spherical particle, near the critical particle diameter for the selected mission, at 7 km/s (15,700 mph).

#### MMOD-IMLI prototype

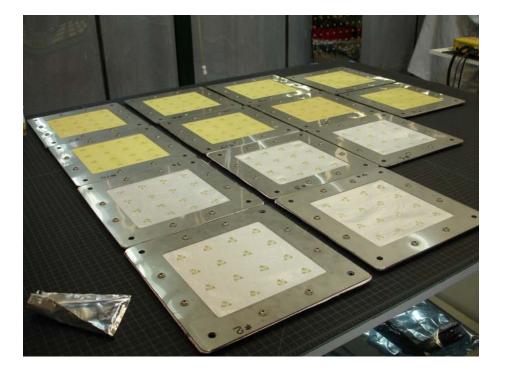






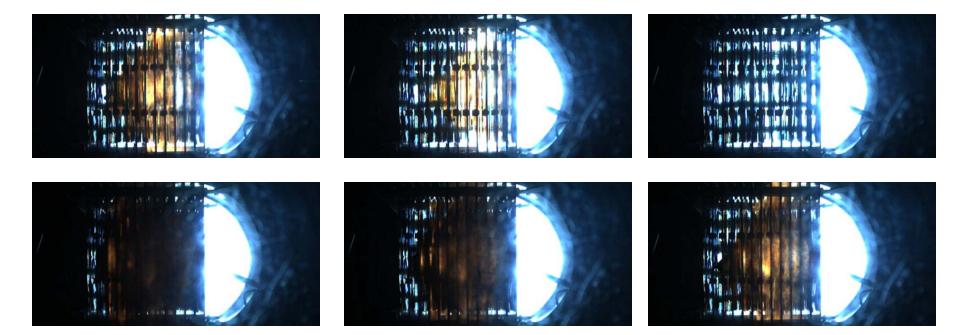
#### **MMOD-IMLI prototype**





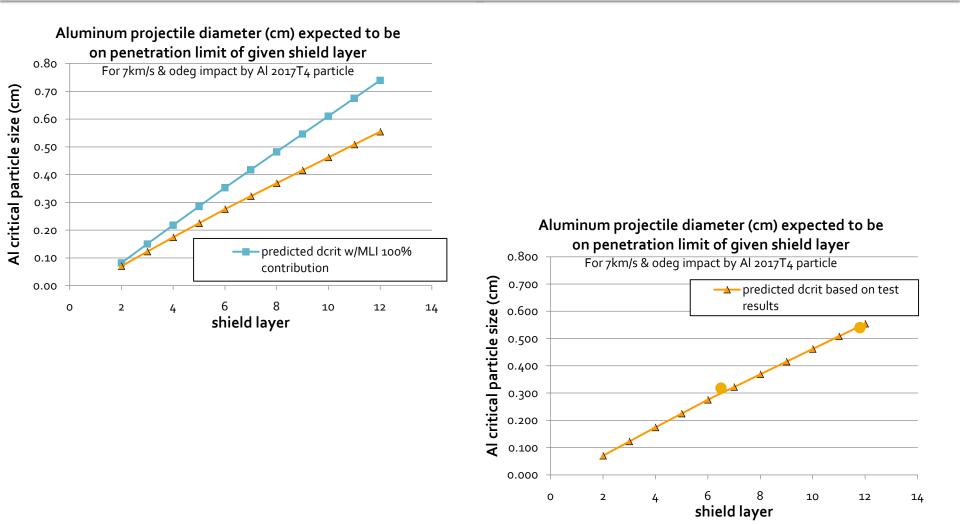
## Hypervelocity Impact Testing





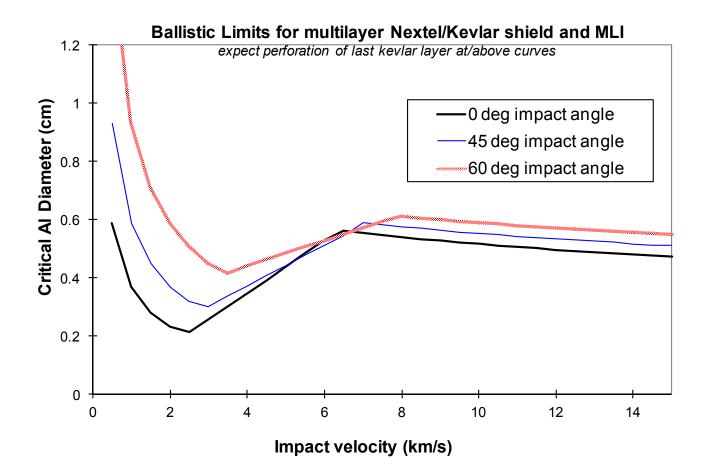
#### **HVI Results**





## **Ballistic Limit Equations**





## **MMOD-IMLI stopping power**



Layer Number	m <sub>sh_eff</sub> (g/cm²)	S (cm)	Aluminum projectile diameter on failure threshold of each shield layer at 7km/s, 0° d (cm)
1	NA	NA	NA
2	0.066	1.8	0.070
3	0.104	3.7	0.123
4	0.141	5.5	0.174
5	0.178	7.4	0.225
6	0.215	9.2	0.275
7	0.248	11.1	0.322
8	0.280	12.9	0.369
9	0.312	14.8	0.415
10	0.345	16.6	0.462
11	0.377	18.5	0.508
12	0.410	20.3	0.555

#### **Ballistic Limit Equations**



$$m_b = 0.185 \, d \cdot \rho_p$$

$$m_w = 29 M_p \cdot V_n \cdot S^{-2} = \frac{29\pi}{6} \cdot d^3 \cdot \rho_p \cdot V_n \cdot S^{-2}$$



- MMOD-IMLI structure performed nearly as modeled
- Completely stopped a 5.4mm particle at 6.6km/s without use of a rear wall
- MMOD-IMLI mass was 8.0 kg/m<sup>2</sup>
- Has a theoretical mass 24% less than advanced multishock shields for same shielding
- Also provides thermal insulation in a single subsystem

#### MMOD thermal testing on cryotank









- 8-layer MMOD-IMLI structure built and installed on 2oL tank
- Heat flux measured via LN2 boiloff calorimetry
- Thermal conductance was 1.58W/m<sup>2</sup>
- Thermal conductivity was 0.12mW/m-K
- IMLI thermal conductivity is 0.066mW/m-K
- Estimated heat flux through full 120-layer MMOD-IMLI structure is 0.10W/m<sup>2</sup>





- Feasibility of MMOD-IMLI was proven, TRL3 achieved
- MMOD-IMLI can provide both high performance thermal insulation and MMOD shielding
- MMOD-IMLI combines thermal barriers, precise layer spacing, and support for high strength ballistic layers
- Thermal performance matches our modeling closely
- MMOD performance can be estimated with BLEs
- MMOD-IMLI can be engineered to meet mission requirements

# Acknowledgements



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- Genevieve Duvaud, Ball Aerospace, MMOD Consultant
- Gary Mills, Ball Aerospace, Staff Engineer
- Phill Tyler, Quest Thermal Group, Mechanical Engineer
- Scott Dye, Quest Thermal Group, Principal Investigator
- Alan Kopelove, Quest Thermal Group, CEO

# Micrometeoroid/Orbital Debris – Integrated Multilayer Insulation

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